

# RD-180



Pratt & Whitney Rocketdyne



**Pratt & Whitney**  
A United Technologies Company

# RD-180

## Propulsion System

Pratt & Whitney Rocketdyne (PWR), through RD AMROSS, is proud to include in its propulsion lineup the RD-180, one of the world's most robust, mission-proven, reliable and powerful kerosene-fueled liquid booster engines in production. By combining liquid oxygen and kerosene propellants, the RD-180 provides clean operation and is one of the most environmentally friendly booster engines in use today.



RD AMROSS, LLC is a partnership of Pratt & Whitney Rocketdyne and NPO Energomash.



The RD-180 is an evolution of the Russian-developed and flight-proven RD-170 engine, which successfully powered numerous Russian missions. So safe and dependable was the RD-170's performance (RD-170 is currently out of production) that it was certified (in Russia) for human flight.

The dual-combustion chambered (fed by a single turbopump) RD-180 utilizes an extremely efficient, high-pressure staged combustion cycle and liquid oxygen-rich preburner, which contributes to the engine's unparalleled performance.

A totally integrated propulsion package, the RD-180 includes hydraulics for control valve actuation and thrust vector gimbaling, pneumatics for valve actuation and system purging, and a thrust frame to distribute loads.

With its initial launch in 2000, the RD-180 became the first Russian rocket engine to power a U.S. expendable launch vehicle. Since then, it has reliably powered many missions for both the Atlas III and V launch vehicles, and currently powers the Atlas V.

Produced in Russia, the RD-180 is provided to the launch industry by RD AMROSS, a partnership between the world's two premier propulsion companies – Pratt & Whitney Rocketdyne of the United States and NPO Energomash of Russia.

### RD-180 Characteristics (100% power)

Nominal Thrust:	(sea level)	860,200 lb
	(vacuum)	933,400 lb
Specific impulse: (sea level)	311.3 sec	
Vacuum specific impulse:	337.8 sec	
Chamber Pressure:	3,722 psia	
Nozzle area ratio:	36.4:1	
Mixture ratio:	2.72	
Length:	140 in.	
Diameter:	124 in.	
Throttle Range:	47% – 100%	
Dry weight:	12,081 lb (5,480 kg)	

### Description

- Staged-combustion cycle engine
- Liquid oxygen/kerosene propellants
- Dual thrust chambers (gimbal +/-8 degrees)
- Single oxygen-rich preburner
- High-pressure turbopump assembly
  - two-stage fuel pump • single-stage oxygen pump • single turbine
- Hypergolic ignition
- Self-contained hydraulic system powered with kerosene from fuel pump
- Minimal interfaces with launch pad and vehicle
- 70% RD-170 parts

### RD-180-Powered Launches

#### Atlas III Launches

- AC-201 May 24, 2000
- AC-204 February 21, 2002
- AC-205 April 12, 2003
- AC-203 December 18, 2003
- AC-202 March 13, 2004
- AC-206 February 3, 2005

#### Atlas V Launches

- AV-001 August 21, 2002
- AV-002 May 13, 2003
- AV-003 July 17, 2003
- AV-005 December 17, 2004
- AV-004 March 11, 2005
- AV-007 August 12, 2005
- AV-010 January 19, 2006
- AV-008 April 20, 2006
- AV-013 March 9, 2007
- AV-009 June 15, 2007
- AV-011 October 11, 2007
- AV-015 December 10, 2007
- AV-006 March 13, 2007
- AV-014 April 14, 2008

PRATT & WHITNEY ROCKETDYNE, a United Technologies company with sites throughout the U.S., is dedicated to providing advanced, reliable, and cost-effective propulsion systems for spacecraft and missile propulsion systems and service.



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